COMMONWEALTH OF AUSTRALIA . DEPARTMENT OF TRANSPORT

SI/784/1505

AIRCRAFT ACCIDENT INVESTIGATION SUMMARY REPORT

1. LOCATION OF OCCURRENCE													
35 Km east of Burra, S.A.					600 fe	Height e.m.s.i. 600 feet approx.		/ 78	0808 h		CST		
2. THE													
Make and Mode	Registratio			Certificate of Airworthiness									
Skycraft Scout			Not app	plicabl	e Not ap	Not applicable							
-Contitionate of Registration issued to Owner:			Operator						Degree of damage to aircraft				
·						1	Destroyed						
	1				Other property damaged								
			<u></u>										
Defects discovered													
Fatigue failure of rudder leading edge tubing.													
3. THE FLIGHT													
Last or intended departure point Time of departure		Next po	oint of inte	nded landing	landing Purpose of fl		ight		Class of operation				
0800 hours		Koomooloo Stat		tation	ion Pleasure)		Private				
4. THE CREW													
	Name	Status	Age	Age Class of		Hours on type	Total hours		Degree of injury		njury		
	Pilot		46	Not a	pplicable	27 / 1	2 7		 Fatal				
		11100	40 Not appi		bbricante	1 1		pprox.					
			<u> </u>				pp2	<u> </u>	<u> </u>				
5. OTHER PERSONS (All passengers and persons injured on ground)													
	Name Status		Degree of Injury			Name		Stat		otus Degree of injury			
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6. RELEVANT EVENTS													
Under certain conditions Air Navigation Order part 95.10 exempts aeroplanes with a maximum all up weight less than 180 kg from normal airworthiness and licencing requirements. The Skycraft Scout is a type of aircraft which falls within this category. During the twelve months that the pilot had owned the aircraft it had been extensively modified. At the time of the accident it was approximately 20 kg heavier than the manufacturer's mmended all up weight. It had however, been fitted with an engine which had a power output some 3.7 kilowatts higher than the original engine. On the day of the accident the weather conditions were fine with light and variable winds. Some 8 minutes after take off the aircraft was observed through binoculars to be cruising normally at about 250 feet altitude when it suddenly descended quickly and disappeared from view. About one hour later, the wreckage was found in flat open terrain. The aircraft had struck the ground in a steep dive killing the pilot. Examination of the wreckage revealed that the leading edge of the rudder had failed due fatigue originating at a bolt hole. This would eventually result in collapsing of the rudder structure.													
7. OPINION AS TO CAUSE													
The cause of the accident was that, following a fatigue failure in the rudder structure, the pilot was deprived of effective control of the aircraft.													
Approved for	. ·				 			D	o 10				
publication			ſ	Delegate of	the Secretary						-		
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