COMMONWEALTH OF AUSTRALIA -AIRCRAFT ACCIDENT INVESTIGATION SUMMARY REPORT

### DEPARTMENT OF TRANSPORT

ublication withis report is authorised by the Secretary under the provisions of Air Navigation Regulations 283 (1)

SI /783/1037

Reference No

I. LOCATION OF OCCURRENCE				
Lilydale, Victoria	Holphi c.m.s.l. 240 feet	Date 14.8.78	Time (Local) 1515 hours	Zene EST
THE AIRCRAFT				

Moke and Model Piper PA-28/140	Registration VH-EGC	Cortificate of Airworthiness Valid from 30.6.76
Certificate of Realistration issued to	Operator	Degree of damage to circreft Substantial
	ć	Other property damaged Fence

#### Defects discovered

- 1. Throttle butterfly dished.
- Throttle shaft bent. 2.
- Inner surfaces of carburettor steel hrottle shaft bushes extremely rough.
- Compression buckling of throttle inner cable.

3. THE FLIGHT						
Last or intended departure point	Time of departure	Next point of intended landing	Purpose of flight	Class of operation		
Mangalore	1430 hours	Lilydale	Pleasure	Private		

#### THE CREW Class of licence Status Hours on type Total hours Degree of injury Ago 38 100 163 Pilot Private Minor

5. OTHER PERSONS (All passengers and persons injured on ground)

Name	Status	Dagree of injury	Name	Status	Degree of injury
	Passenger	Serious			,

# 6. RELEVANT EVENTS

The purpose of the flight was for the pilot to gain some recent experience for the renewal of his private licence. While returning from a flight to Mangalore the throttle control became stiff to operate and appeared to be restricted in its movement. On arriving back at Lilydale the pilot planned for and made a steep approach to land in case the throttle control malfunctioned. On final approach, when he believed the aircraft was overshooting, he closed the throttle and selected flaps full down. Later on final approach he assessed that the aircraft was undershooting and on attempting to apply power he found that the throttle was jammed closed. He raised the flaps in an attempt to increase the gliding range but the aircraft undershot the end of the airstrip and heavily contacted e embankment of a drain just inside the airfield boundary. The landing gear collapsed and the ircraft slid some 20 metres before stopping.

The aircraft's engine and carburettor had been overhauled some 5.85 flying hours prior to the accident. At this time a new throttle butterfly valve, throttle shaft and throttle shaft bushes were installed. Post accident inspection of the carburettor indicated that the steel throttle shaft bushes had been binding on the steel throttle shaft. During the investigation samples of new butterfly valves, throttle shafts and throttle shaft bushes were obtained. All had defects approximately similar to those listed in section 2 above.

## 7. OPINION AS TO CAUSE

The cause of the accident was that following a misjudged landing approach the pilot was unable to apply engine power due to jamming in the throttle system. The probable cause of the throttle system jamming was seizure of the throttle shaft in the throttle shaft bushes.

Approved for		Date
publication	Delegate of the Socratory	



# **MINUTE**

SIGS (B)

It review of the refort at folio 9 has been confleted, and a draft summary refort frequested or requested

The reford concerns a defective carburethor throttle shaft assembly which 's eiged', depriving the filed of throttle control at a critical stage of flight. The filed became awars of a throttle problem while authorne and planned a glide approach should the throttle fam. Due to his lack of continuity in flying practice however, he misfudged the approach. When he alterifted to open the throttle to corred an undershoot, the throttle jammed.

followed up, itsede from the airworthings aspects, there is a lesson to be learned from the accident. Pelots of lingle engined aircraft have a need to fractice forced landings just as much as much engineed protest much fractice asymmetric aircraft handling.

. your attention is drawn to the requests contained in the covering memo.

78-2-79

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2. THE AIRCRAFT									
Make and Model		Registration		Certificate of Airworthiness					
Piper PA-28/140		VH-EGO	3	Valid	d from 30.	6.76			
Certificate of Registration issued to		Operator				Degree of damage to circustra Substantial			
						Other property damaged			
Defects discovered				<del></del>		Fence		<del></del>	
Carburettor throttle shat Compression failure of the	_		e shaft	bushes.		÷	8		
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3. THE FLIGHT						<del></del>	<del></del>		
Last or intended departure point	Time of departure	Next po	int of intende	ed tanding	Purpose of fl	light Class of operation			
Mangalor <del>e</del>	1430 hours	Lilydale		ıle	Pleas	Pleasure		Private	
4. THE CREW									
Name	Status	Age	Class	of licence	Hours on type	Total hours	D.	gree of in	jury
	Pilot	38	Pr	ivate	100	163 Minor		Minor	
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	1					1			
5. OTHER PERSONS (All pass	engers and person	is injured o	n ground)	<del></del>	<u></u>	<del></del>			
Name	Status	Degree of			Name	S	tatus	Degree	of injury
	Passangan	Soni	2336						
	Passenger	Delit	Serious				1		
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6. RELEVANT EVENTS	<del></del>				<del>- · · · · · · · · · · · · · · · · · · ·</del>				
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attempting to apply power	he found tha	t the th	rottle w	ing inme	d closed.	He raised	the fla	ns to	
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# rease the gliding range but the aircraft undershot the end of the airstrip and landed heavily the embankment of a drain just inside the airfield boundary. The landing gear collapsed and the aircraft slid some 20 metres before stopping.

# 7. OPINION AS TO CAUSE

The cause of the accident was that, on becoming aware of a throttle control malfunction, the pilot did not use the best operating techniques to ensure a safe landing.

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