Aviation Safety Investigation Report 199604069

Cessna Aircraft Company Centurion

07 December 1996

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NOTE: All air safety occurrences reported to the ATSB are categorised and recorded. For a detailed explanation on Category definitions please refer to the ATSB website at www.atsb.gov.au.

199604069

The Bureau did not conduct an on scene investigation of this occurrence. The information presented below was obtained from information supplied to the Bureau.

Occurrence Number: 199604069 Occurrence Type: Accident

Location: Darwin, Aerodrome

State: NT Inv Category: 4

Date: Saturday 07 December 1996

Time: 1309 hours **Time Zone** CST

Highest Injury Level: None

Aircraft Manufacturer: Cessna Aircraft Company

Aircraft Model: 210L

Aircraft Registration: VH-TIJ Serial Number: 21060591

Type of Operation: Charter Passenger

Damage to Aircraft: Substantial **Departure Point:** Bamyili NT

Departure Time:

Destination: Hodgson River Station NT

Crew Details:

	Hours on		
Role	Class of Licence	Type Hours Total	
Pilot-In-Command	Commercial	500.0	800

Approved for Release: Friday, December 20, 1996

Arriving at the destination the landing gear failed to extend when the pilot selected it down.

His attempts to extend it using the manual hand pump were unsuccessful as he was unable to build up enough hydraulic pressure in the system to operate the gear. Hydraulic fluid was then noticed seeping from behind the centre console which he identified as a serious leak from the power pak. He was able to replenished the lost hydraulic fluid with spare engine oil carried onboard the aircraft, but this failed to rectify the loss of hydraulic pressure.

After calculating the aircraft's endurance the pilot decided to return to Darwin where he carried out a wheels up landing.

An inspection revealed that one of the "O" ring seals located at the joint between the landing gear manifold and the power pak was split, causing an hydraulic fluid leak, and loss of hydraulic pressure.

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The manifold had been removed the previous day to change the servo-valve, and it is possible the "O" ring was damaged during reinstallation. Several successful gear retraction tests had been carried out after completion of the work, followed by about 9 gear cycles in flight before the leak developed.