Aviation Safety Investigation Report 199702089

Saab Aircraft AB 340

27 June 1997

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Investigations commenced on or before 30 June 2003, including the publication of reports as a result of those investigations, are authorised by the Executive Director of the Bureau in accordance with Part 2A of the Air Navigation Act 1920.

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NOTE: All air safety occurrences reported to the ATSB are categorised and recorded. For a detailed explanation on Category definitions please refer to the ATSB website at www.atsb.gov.au.

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The Bureau did not conduct an on scene investigation of this occurrence. The information presented below was obtained from information supplied to the Bureau.

Occurrence Number: 199702089 Occurrence Type: Incident

Location: Sydney, Aerodrome

NSW State: **Inv Category:**

Date: Friday 27 June 1997

Time: Time Zone **EST** 1116 hours

Highest Injury Level: None

Aircraft Manufacturer: Saab Aircraft AB

Aircraft Model: SF-340B

Aircraft Registration: VH-CMH Serial Number: 327

Type of Operation: Air Transport Domestic Low Capacity Passenger Scheduled

Damage to Aircraft: Minor

Departure Point: Sydney NSW 1116 EST **Departure Time:**

Destination: Wagga Wagga NSW

Approved for Release: Thursday, September 25, 1997

The SAAB 340 departed normally from runway 16R, but shortly after becoming airborne the crew requested a return to land. When asked if operations were normal the crew reported the aircraft had suffered an engine failure. ATC activated the crash alarm and closed the airport. The crew were offered a circuit to either runway 07 or 34L/R, but elected to continue for a right circuit to runway 16R. The aircraft was flown on a visual low level circuit to a safe landing. Preliminary investigation revealed major damage to the power turbine section of the left engine.

The engine was subsequently returned to the manufacturers engine overhaul facility for examination. Their engineering report revealed that the stage two turbine forward cooling plate had separated at the outer rim. The report noted that the failure was consistent with other cooling plate separation events.

The manufacturer had previously issued a service bulletin which introduces improvements to the cooling plate to preclude this type of failure. The operator had a program to incorporate this modification during planned hot section module changes. Incorporation of the modification to the failed engine was planned for the week following the incident.