COMMONWEALTH OF AUSTRALIA
DEPARTMENT OF CIVIL AVIATION

AIRCRAFT ACCIDENT INVESTIGATION SUMMARY REPORT

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AS/691/1027

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1. LOCATION OF OCCURRE	NCE					· · · · · · · · · · · · · · · · · · ·			
				Height a.r	1	Date			Zone
22 miles south west of M	undubbera, Que	ensland.		1000	feet	5.10.69 1250 ES			EST
3. THE AIRCRAFT		· · · · · · ·		· · ·	·	•			
Make and Model		Contined of			Valid to				
Meta Sokol L40		VH-DU	<u>r</u>	Airv	vorthiness	28,1,66	3.1.66 27.1.75		
Registered Owner		Operator			1	Degree of damage to aircraft			
	•				Substantial				
	}				Other property damaged				
						Nil			
Defects discovered									•
Fuel injection priming pi	pe to No. 3 cyli	nder crac	ked nea	ar injector	pump end.				
Washers on all priming n	•			,					
Oil seal for starboard ma	•	-	•						
									
3. THE FLIGHT									
Last or intended departure point	Time of departure	Next point of intended landi		ded landing	ling Purpose of flight		Class of operation		
Roma	1132	Maryborough		ıgh	Air Racing		Private		
			-J						
THE CREW	·								
Name .	Status	Age	Class of licence		Hours on type	Total hours	Degree of i		njury
		1							
	Pilot	46	Com	mercial	40	1200	Nil		
					1				
5. OTHER PERSONS (All pas	sengers and person	s injured o	n ground)			-	 		
Name	Status	Degree of	Degree of injury		Name		latus	Degree	of injury
	Doggongon	3721	1						
	Passenger		Nil						
	Passenger	Nil							
. CONTRAVENTIONS OF RE	EGULATIONS AND	ORDERS						•	
Regulation or Order No.				Nature of	ontravention				
		,					. e e		

RELEVANT EVENTS

Whilst cruising in level flight the engine, without warning, commenced to vibrate severely. Engine instrumentation remained normal and a magneto check was satisfactory. The vibration increased rapidly to such a degree that the pilot had no alternative but to close the throttle fully and carry out a forced landing. The aircraft was over very rough terrain and the only field available was 1,580 feet in length with trees to 70 feet high at one end and to 20 feet high at the other. The approach was made towards the taller timber, with the wheels down and half flap in a cross wind of about 10 knots from the left. After touching down the pilot realized that the aircraft would overrun the available area and he immediately initiated a turn to the right. Because of the speed and the rough surface, the aircraft skidded sideways and the undercarriage collapsed.

The aircraft performance chart indicates that an actual distance of 1,740 feet is required to stop the aircraft from a height of 50 feet when using full flap under the conditions existing at the time of the landing.

8. OPINION AS TO CAUSE

The cause of the accident was that following an engine malfunction the pilot was compelled to land the aircraft on an unsuitable area. The engine malfunction was probably the result of defects in the fuel injection system.

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(D.S. GRAHAM)

Designation
Assistant Director-General
(Air Safety Investigation)

Date 4.2.1971