Aviation Safety Investigation Report 198803423

Piper PA-24

12 January 1988

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Investigations commenced on or before 30 June 2003, including the publication of reports as a result of those investigations, are authorised by the CEO of the Bureau in accordance with Part 2A of the Air Navigation Act 1920.

Investigations commenced after 1 July 2003, including the publication of reports as a result of those investigations, are authorised by the CEO of the Bureau in accordance with the Transport Safety Investigation Act 2003 (TSI Act). Reports released under the TSI Act are not admissible as evidence in any civil or criminal proceedings.

NOTE: All air safety occurrences reported to the ATSB are categorised and recorded. For a detailed explanation on Category definitions please refer to the ATSB website at <u>www.atsb.gov.au</u>.

Occurrence Number: Location: Date: Highest Injury Level: Injuries:		St George QLD 12 January 1988			Occurrence Type: Accident Time: 1540	
injuites.			Fatal	Serious	Minor	None
	Cre	ew	0	0	1	1
	Gr	ound	0	0	0	-
	Pas	ssenger	0	0	0	1
	То	tal	0	0	0	2
A : 64 D-4-11		4				
Aircraft Details:	±					
Registration:	VH-COM					
Serial Number:	24-97					
Operation Type:	Private (Business)					
Damage Level:	Substantial					
Departure Point:	"Lea Wah" (Near Bollon) QLD					
Departure Time:	1516					
Destination:	Toowoomba QLD					

Approved for Release: March 16th 1989

Circumstances:

The aircraft was cruising normally at 7000 feet when the engine failed without warning. The pilot carried out checks and attempted to restart the engine without success. During the ensuing forced landing the pilot prematurely turned off the master switch, therefore, the wheels did not extend when he selected gear down. The engine failure was caused by the separation of crankshaft idler gear teeth due to high cycle low stress fatigue which had initiated from pre-existing overload cracks at the base of the teeth.

Significant Factors:

It was considered that the following factors were relevant to the development of the accident:

1. The idler gear failure was considered to be as a result of the previous unusually high load causing cracking at the base of the teeth. The cause of the overload is not known.

2. The pilot had assessed the field that he had chosen as probably suitable for a normal landing. He delayed gear selection until he was sure of making a satisfactory approach and to better assess the surface.

3. The gear did not extend when selected down on final because the pilot had completed his final checks on downwind which included turning off the electrical services master switch.

4. The pilot had insufficient time whilst on final approach to assess why the gear had failed to extend and chose to concentrate on carrying out a successful wheels up landing.