1

Aviation Safety Investigation Report 198900826

Bell 47G-3B1

15 September 1989

Readers are advised that the Australian Transport Safety Bureau investigates for the sole purpose of enhancing transport safety. Consequently, Bureau reports are confined to matters of safety significance and may be misleading if used for any other purposes.

Investigations commenced on or before 30 June 2003, including the publication of reports as a result of those investigations, are authorised by the CEO of the Bureau in accordance with Part 2A of the Air Navigation Act 1920.

Investigations commenced after 1 July 2003, including the publication of reports as a result of those investigations, are authorised by the CEO of the Bureau in accordance with the Transport Safety Investigation Act 2003 (TSI Act). Reports released under the TSI Act are not admissible as evidence in any civil or criminal proceedings.

NOTE: All air safety occurrences reported to the ATSB are categorised and recorded. For a detailed explanation on Category definitions please refer to the ATSB website at www.atsb.gov.au.

Occurrence Number: 198900826 Occurrence Type: Accident

Location: Katherine Gorge NT

Date: 15 September 1989 **Time:** 1815

Highest Injury Level: Nil

Injuries:

	Fatal	Serious	Minor	None
Crew	0	0	1	1
Ground	0	0	0	-
Passenger	0	0	0	2
Total	0	0	0	3

Aircraft Details: Bell 47G-3B1 **Registration:** VH-HMU **Serial Number:** 3392

Operation Type: Aerial Work **Damage Level:** Substantial

Departure Point: Katherine Gorge NT

Departure Time: 1815

Destination: Katherine Gorge NT

Approved for Release: 15th November 1989

Circumstances:

The helicopter took off and entered a hover in ground effect before a climbing turn to the left was initiated from approximately 10 feet above ground level(agl). Immediately the turn was commenced the helicopter started to lose mainrotor rpm. The terrain drops away from the heli-pad and the helicopter was now 35 to 40 feet agl. The pilot attempted to regain rotor rpm by lowering the collective and increasing throttle, however, RPM was not regained and the helicopter continued in the descent until ground impact. The pilot was interviewed by an Examiner of Airmen who was of the opinion that the symptoms described by the pilot indicated that the mainrotor was overpitched during the climbing turn.